

# Computational Astrophysics 5

## Higher-order and AMR schemes

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# Outline

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- The Godunov Method
- Second-order scheme with MUSCL
- Slope limiters and TVD schemes
- Characteristics tracing and 2D slopes.
- Adaptive Mesh Refinement

# Godunov scheme for hyperbolic systems

The system of conservation laws

$$\partial_t \mathbf{U} + \partial_x \mathbf{F} = 0$$

is discretized using the following integral form:

$$\frac{\mathbf{U}_i^{n+1} - \mathbf{U}_i^n}{\Delta t} + \frac{\mathbf{F}_{i+1/2}^{n+1/2} - \mathbf{F}_{i-1/2}^{n+1/2}}{\Delta x} = 0$$

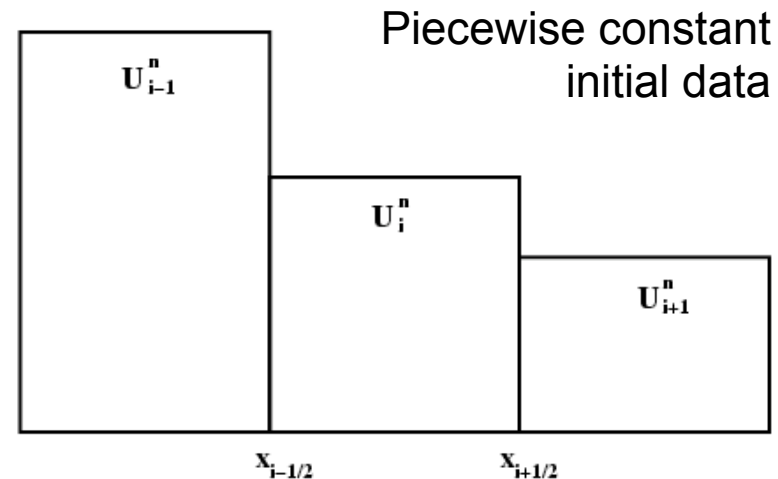
The time average flux function is computed using the self-similar solution of the inter-cell Riemann problem:

$$\mathbf{U}_{i+1/2}^*(x/t) = \mathcal{RP}[\mathbf{U}_i^n, \mathbf{U}_{i+1}^n]$$

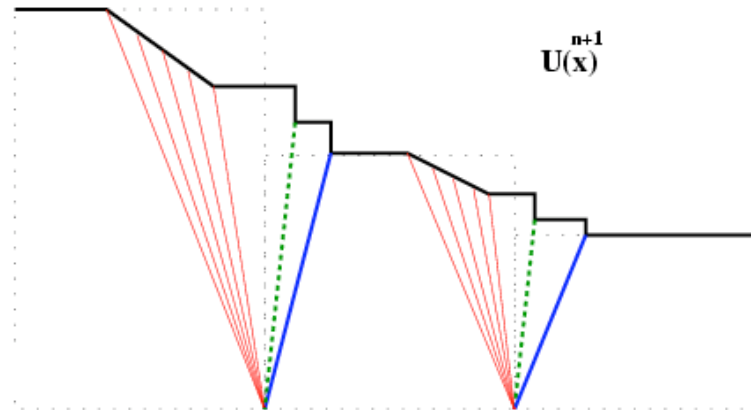
$$\mathbf{F}_{i+1/2}^{n+1/2} = \mathbf{F}(\mathbf{U}_{i+1/2}^*(0))$$

This defines the Godunov flux:

$$\mathbf{F}_{i+1/2}^{n+1/2} = \mathbf{F}^*(\mathbf{U}_i^n, \mathbf{U}_{i+1}^n)$$



- Godunov, S. K. (1959), A Difference Scheme for Numerical Solution of Discontinuous Solution of Hydrodynamic Equations, *Math. Sbornik*, **47**, 271-306, translated US Joint Publ. Res. Service, JPRS 7226, 1969.



Advection: 1 wave, Euler: 3 waves, MHD: 7 waves

## Higher Order Godunov schemes

Godunov method is stable but very diffusive. It was abandoned for two decades, until...

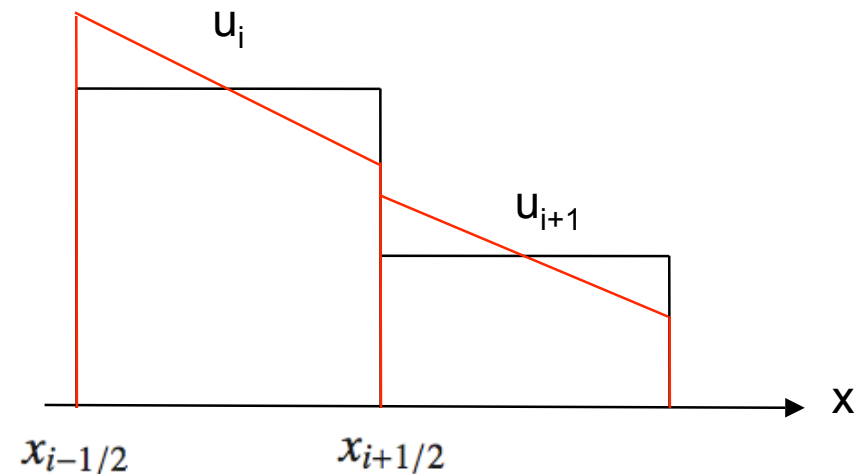


Bram Van Leer

- **van Leer, B.** (1979), Towards the Ultimate Conservative Difference Scheme, V. A Second Order Sequel to Godunov's Method, *J. Com. Phys.*, 32, 101–136.

## Second Order Godunov scheme

Piecewise linear approximation of the solution:



The linear profile introduces a length scale: the Riemann solution is not self-similar anymore:

$$\mathbf{F}_{i+1/2}^{n+1/2} \neq \mathbf{F}(\mathbf{U}_{i+1/2}^*(0))$$

The flux function is approximated using a *predictor-corrector* scheme:

$$\mathbf{F}_{i+1/2}^{n+1/2} = \frac{1}{\Delta t} \int_{t^n}^{t^{n+1}} \mathbf{F}(x_{i+1/2}, t) dt \longrightarrow \mathbf{F}_{i+1/2}^{n+1/2} \simeq \mathbf{F}(\mathbf{U}_{i+1/2}^*(\frac{\Delta t}{2}))$$

The *corrected* Riemann solver has now *predicted* states as initial data:

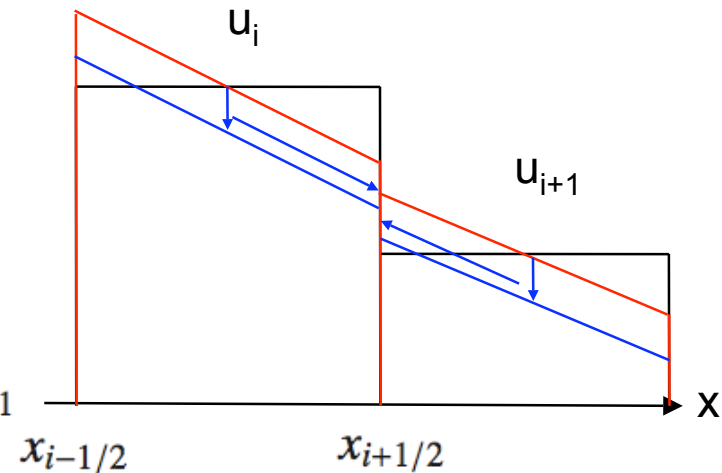
$$\mathbf{U}_{i+1/2}^*(x/t) = \mathcal{RP}[\mathbf{U}_{i+1/2,L}^{n+1/2}, \mathbf{U}_{i+1/2,R}^{n+1/2}]$$

## Predictor Step for the advection equation

The predicted states are computed using a Taylor expansion in space and time:

$$u_{i+1/2,L}^{n+1/2} = u_i^n + \frac{\Delta t}{2} \left( \frac{\partial u}{\partial t} \right)_i + \frac{\Delta x}{2} \left( \frac{\partial u}{\partial x} \right)_i$$

$$u_{i+1/2,R}^{n+1/2} = u_{i+1}^n + \frac{\Delta t}{2} \left( \frac{\partial u}{\partial t} \right)_{i+1} - \frac{\Delta x}{2} \left( \frac{\partial u}{\partial x} \right)_{i+1}$$



Second order predicted states are the new initial conditions for the Riemann solver:

$$u_{i+1/2,L}^{n+1/2} = u_i^n + (1 - C) \frac{\Delta x}{2} \left( \frac{\partial u}{\partial x} \right)_i \quad u_{i+1/2,R}^{n+1/2} = u_{i+1}^n - (1 + C) \frac{\Delta x}{2} \left( \frac{\partial u}{\partial x} \right)_{i+1}$$

The *corrected* flux function is the *upwind* predicted state:

$$f_{i+1/2}^{n+1/2} = au_{i+1/2,L}^{n+1/2} \quad \text{if } a > 0 \quad f_{i+1/2}^{n+1/2} = au_{i+1/2,R}^{n+1/2} \quad \text{if } a < 0$$

## Modified equation for the second order scheme

$$\frac{u_i^{n+1} - u_i^n}{\Delta t} + a \frac{u_i^n - u_{i-1}^n}{\Delta x} + \frac{a}{2}(1 - C) \left[ \left( \frac{\partial u}{\partial x} \right)_i - \left( \frac{\partial u}{\partial x} \right)_{i-1} \right] = 0$$

Taylor expansion in space and time up to third order:

$$\begin{aligned} u_i^{n+1} &= u_i^n + \Delta t \left( \frac{\partial u}{\partial t} \right) + \frac{(\Delta t)^2}{2} \left( \frac{\partial^2 u}{\partial t^2} \right) + \frac{(\Delta t)^3}{6} \left( \frac{\partial^3 u}{\partial t^3} \right) \\ u_{i-1}^n &= u_i^n - \Delta x \left( \frac{\partial u}{\partial x} \right) + \frac{(\Delta x)^2}{2} \left( \frac{\partial^2 u}{\partial x^2} \right) - \frac{(\Delta x)^3}{6} \left( \frac{\partial^3 u}{\partial x^3} \right) \\ \left( \frac{\partial u}{\partial x} \right)_{i-1} &= \left( \frac{\partial u}{\partial x} \right)_i - \Delta x \left( \frac{\partial^2 u}{\partial x^2} \right) + \frac{(\Delta x)^2}{2} \left( \frac{\partial^3 u}{\partial x^3} \right) \end{aligned}$$

We obtain a *dispersive term* as leading-order error.

Von Neumann analysis says the scheme is stable for  $C < 1$ .

$$\left( \frac{\partial u}{\partial t} \right) + a \left( \frac{\partial u}{\partial x} \right) = a \frac{\Delta x^2}{6} (1 - C) \left( \frac{1}{2} - C \right) \left( \frac{\partial^3 u}{\partial x^3} \right) + O(\Delta t^3, \Delta x^3)$$

## Summary: the MUSCL scheme for systems

Compute second order predicted states using a Taylor expansion:

$$\left\{ \begin{array}{l} \mathbf{W}_{i+1/2,L}^{n+1/2} = \mathbf{W}_i^n + \frac{\Delta t}{2} \left( \frac{\partial \mathbf{W}}{\partial t} \right)_i + \frac{\Delta x}{2} \left( \frac{\partial \mathbf{W}}{\partial x} \right)_i \\ \mathbf{W}_{i+1/2,R}^{n+1/2} = \mathbf{W}_{i+1}^n + \frac{\Delta t}{2} \left( \frac{\partial \mathbf{W}}{\partial t} \right)_{i+1} - \frac{\Delta x}{2} \left( \frac{\partial \mathbf{W}}{\partial x} \right)_{i+1} \end{array} \right.$$

$$\left\{ \begin{array}{l} \mathbf{W}_{i+1/2,L}^{n+1/2} = \mathbf{W}_i^n + (\mathbf{I} - \mathbf{A} \frac{\Delta t}{\Delta x}) \frac{\Delta x}{2} \left( \frac{\partial \mathbf{W}}{\partial x} \right)_i \\ \mathbf{W}_{i+1/2,R}^{n+1/2} = \mathbf{W}_{i+1}^n - (\mathbf{I} + \mathbf{A} \frac{\Delta t}{\Delta x}) \frac{\Delta x}{2} \left( \frac{\partial \mathbf{W}}{\partial x} \right)_{i+1} \end{array} \right.$$

Update conservative variables using corrected Godunov fluxes

$$\mathbf{F}_{i+1/2}^{n+1/2} = \mathbf{F}^*(\mathbf{W}_{i+1/2,L}^{n+1/2}, \mathbf{W}_{i+1/2,R}^{n+1/2}) \quad \frac{\mathbf{U}_i^{n+1} - \mathbf{U}_i^n}{\Delta t} + \frac{\mathbf{F}_{i+1/2}^{n+1/2} - \mathbf{F}_{i-1/2}^{n+1/2}}{\Delta x} = 0$$



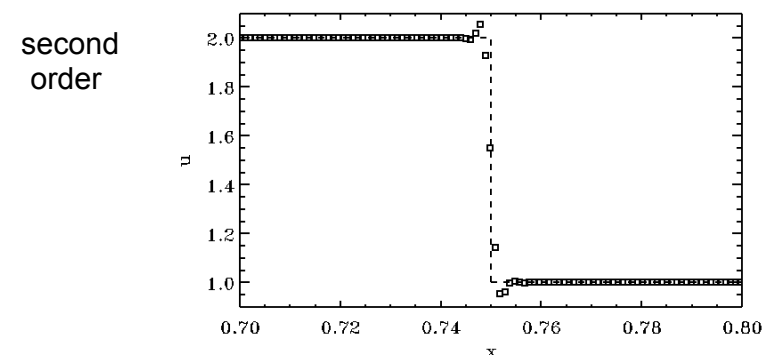
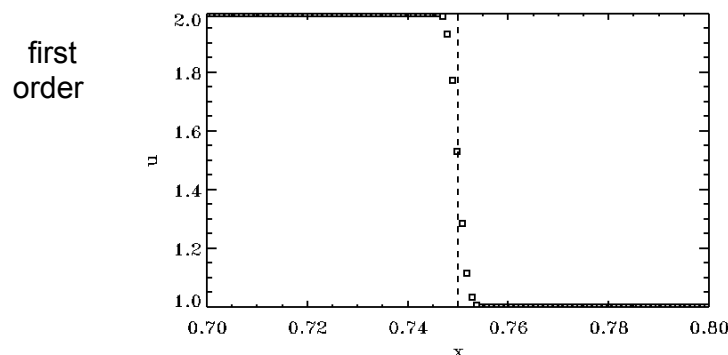
# Monotonicity preserving schemes

We use the central finite difference approximation for the slope:

$$\left(\frac{\partial u}{\partial x}\right)_i = \frac{\Delta u_i}{\Delta x} = \frac{u_{i+1} - u_{i-1}}{2}$$

Second order linear scheme.

In this case, the solution is oscillatory, and therefore non physical.



Oscillations are due to the *non monotonicity* of the numerical scheme.

A scheme is monotonicity preserving if:

- No new local extrema are created in the solution
- Local minimum (maximum) non decreasing (increasing) function of time.

**Godunov theorem:** only first order linear schemes are monotonicity preserving !

## Slope limiters

Harten introduced the Total Variation of the numerical solution:

$$TV^n = \sum_i^n |u_{i+1} - u_i|$$

**Harten's theorem:** a Total Variation Diminishing (TVD) scheme is monotonicity preserving.

$$TV^{n+1} \leq TV^n$$

Design non-linear TVD second order scheme using slope limiters:

$$\left(\frac{\partial u}{\partial x}\right)_i = \frac{\Delta u_i}{\Delta x} = \lim(u_{i-1}, u_i, u_{i+1}) \left(\frac{u_{i+1} - u_{i-1}}{2}\right)$$

where the slope limiter is a non-linear function satisfying:

$$0 \leq \lim(u_{i-1}, u_i, u_{i+1}) \leq 1$$

- Harten, Ami (1983), "High resolution schemes for hyperbolic conservation laws", *J. Comput. Phys* **49**: 357-393, [doi:10.1006/jcph.1997.5713](https://doi.org/10.1006/jcph.1997.5713)

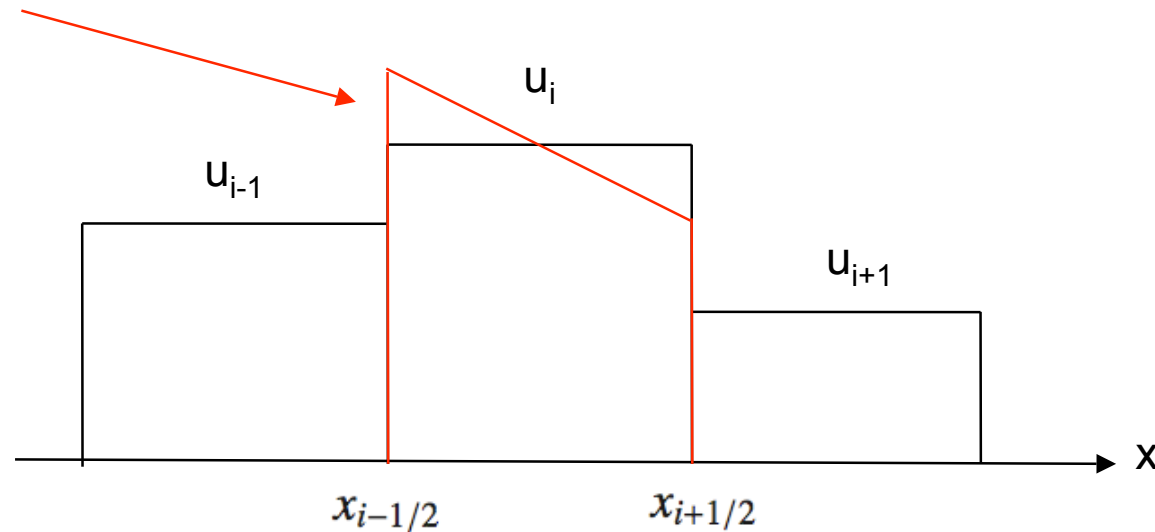
## No local extrema

$$\left(\frac{\partial u}{\partial x}\right)_i = \frac{\Delta u_i}{\Delta x} = \lim(u_{i-1}, u_i, u_{i+1}) \left(\frac{u_{i+1} - u_{i-1}}{2}\right)$$

We define 3 local slopes: left, right and central slopes

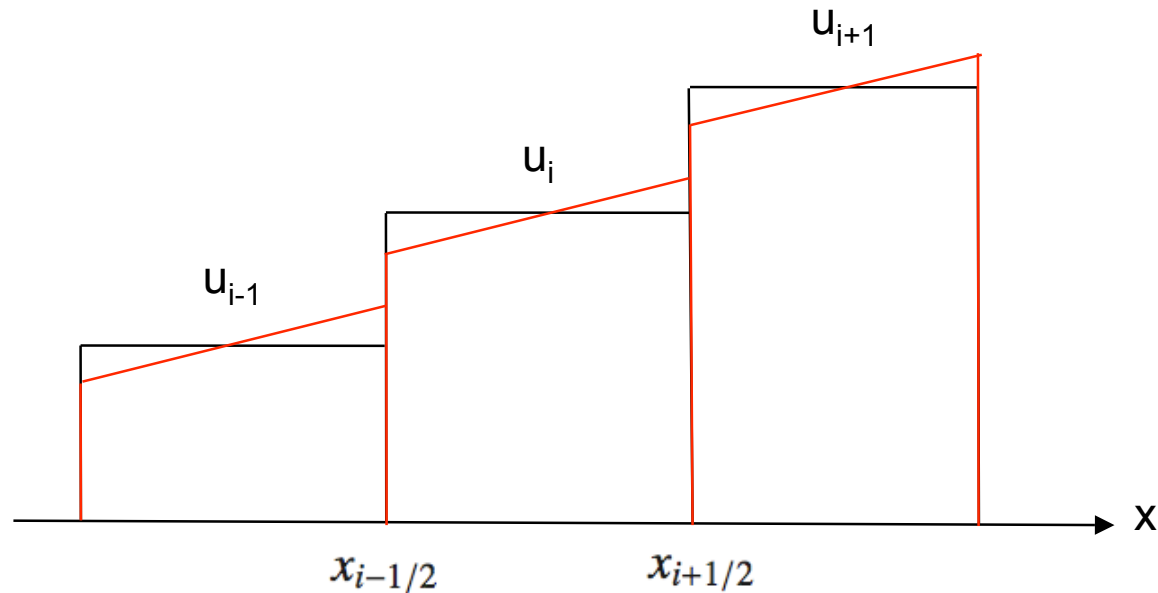
$$\Delta u_L = u_i - u_{i-1} \quad \Delta u_R = u_{i+1} - u_i \quad \text{and} \quad \Delta u_C = \frac{u_{i+1} - u_{i-1}}{2}$$

New maximum !



**For all slope limiters:**  $\Delta u_i = 0$  if  $\Delta u_L \Delta u_R < 0$

## The *minmod* slope



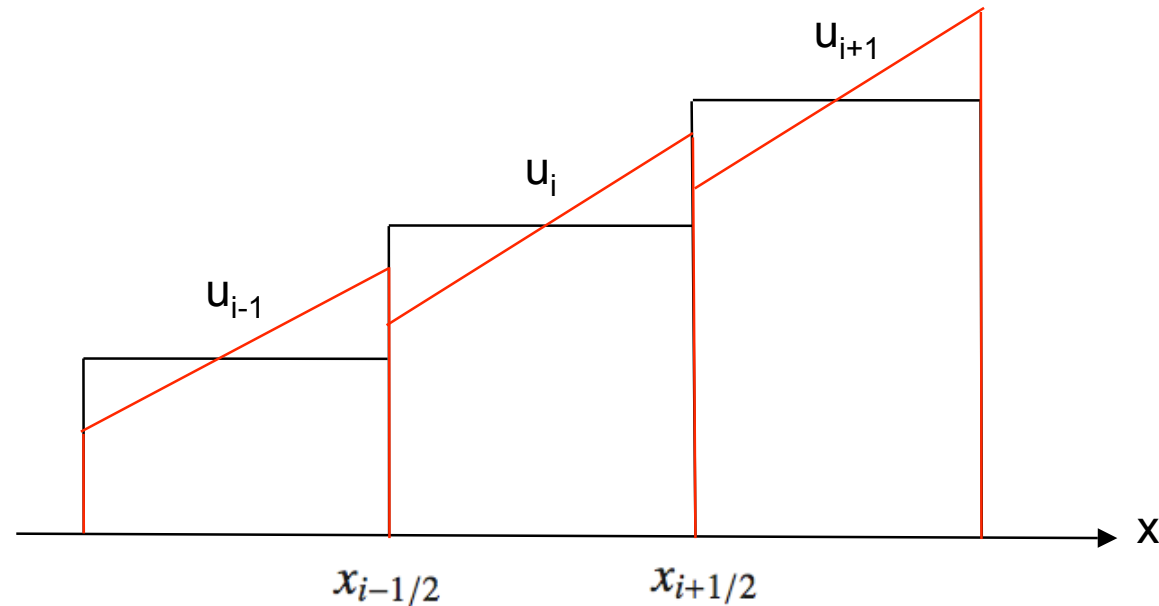
Linear reconstruction is monotone at time  $t^n$

$$u_{i+1/2,L}^n = u_i^n + \frac{\Delta u_i}{2} \quad u_{i-1/2,R}^n = u_i^n - \frac{\Delta u_i}{2}$$

Minmod slope limiting is never truly second order !

$$u_{i+1/2,L}^n \leq u_{i+1/2,R}^n \quad \Delta u_i = \min(\Delta u_L, \Delta u_R)$$

## The moncen slope



Extreme values must be bounded by the *initial average* states.

$$u_{i-1/2,R}^n = u_i^n - \frac{\Delta u_i}{2}$$

$$u_{i+1/2,L}^n = u_i^n + \frac{\Delta u_i}{2}$$

$$u_{i-1}^n \leq u_{i-1/2,R}^n \leq u_i^n$$

$$u_i^n \leq u_{i+1/2,L}^n \leq u_{i+1}^n$$

$$\Delta u_i = \min(2\Delta u_L, \Delta u_C, 2\Delta u_R)$$

## The *superbee* slope

Predicted states must be bounded by the initial average states.

$$u_{i+1/2,L}^{n+1/2} = u_i^n + (1 - C) \frac{\Delta u_i}{2}$$

$$u_{i+1/2,R}^{n+1/2} = u_{i+1}^n - (1 + C) \frac{\Delta u_{i+1}}{2}$$

TVD constraint is preserved by the Riemann solver.

$$u_i^n \leq u_{i+1/2,L}^{n+1/2} \leq u_{i+1}^n$$

$$u_{i-1}^n \leq u_{i-1/2,R}^{n+1/2} \leq u_i^n$$

The Courant factor now enters the slope definition.

$$\Delta u_i = \min\left(\frac{2}{1 + C} \Delta u_L, \frac{2}{1 - C} \Delta u_R\right)$$

## *The ultrabee slope*

Use the final state to compute the slope limiter.

$$u_i^{n+1} = u_i^n(1 - C) + u_{i-1}^n C - \frac{C}{2}(1 - C)(\Delta u_i - \Delta u_{i-1}) = 0$$

Upwind Total Variation constraint.

$$u_{i-1}^n \leq u_i^{n+1} \leq u_i^n$$

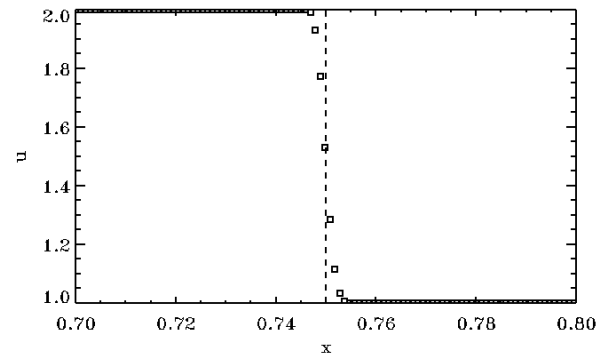
Strict Total Variation preserving limiter.

$$\text{if } C > 0 \quad \Delta u_i = \min\left(\frac{2}{C}\Delta u_L, \frac{2}{1-C}\Delta u_R\right)$$

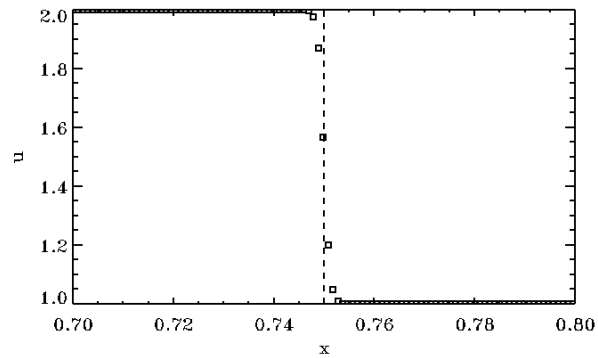
$$\text{if } C < 0 \quad \Delta u_i = \min\left(\frac{2}{1+C}\Delta u_L, \frac{2}{-C}\Delta u_R\right)$$

# Summary: slope limiters

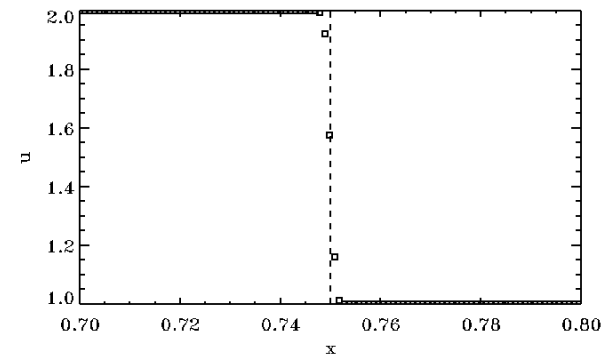
first order



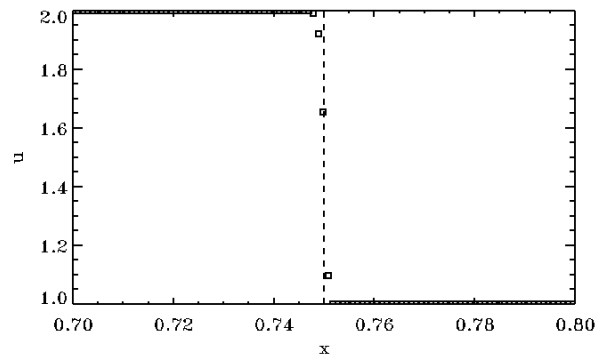
minmod



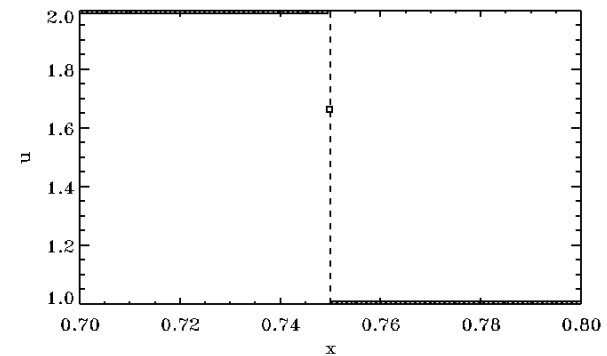
moncen



superbee



ultrabee

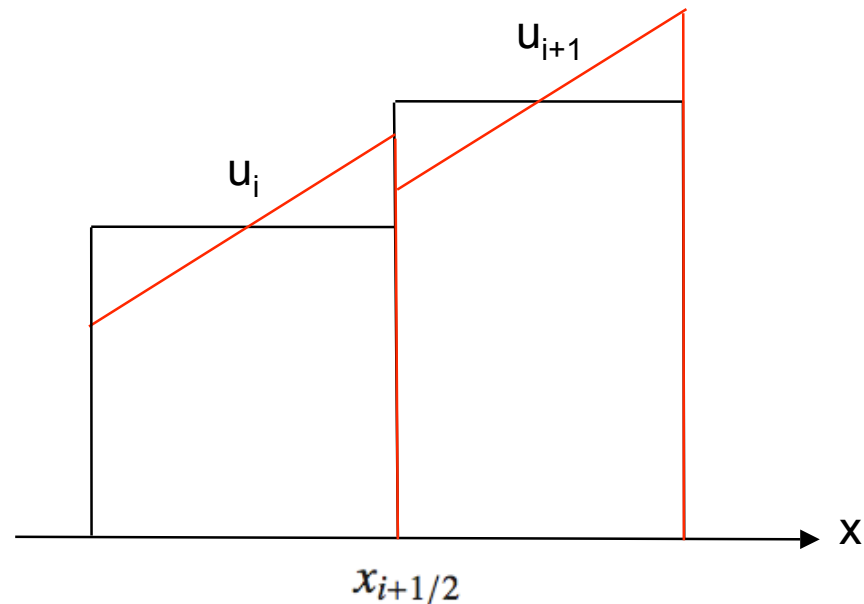




## Summary: slope limiters

The previous analysis is valid only for the advection equation.

Non-linear systems: the wave speeds depend on the initial states (L and R).

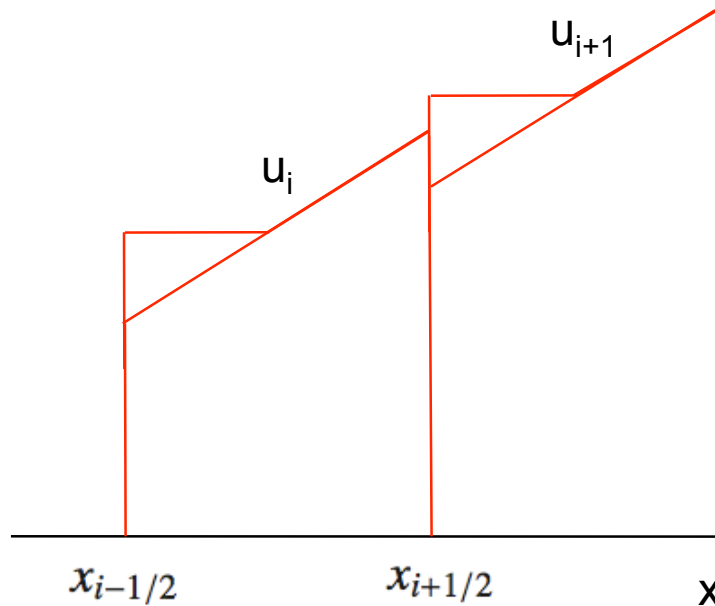


MinMod is the only monotone slope limiter before the Riemann solver !

Superbee and Ultrabee must not be used for non-linear systems !

MonCen can be used, but with care: the characteristics tracing method.

## Non-linear systems: characteristics tracing.



Non-linear Riemann problems: waves speeds depend on the input states.

TVD schemes are not necessary monotone.

Modify the predictor step according to the local Riemann solution: Piecewise Linear Method (PLM) and Piecewise Parabolic Method (PPM).

$$\begin{aligned} \text{If } (C_k)_i > 0 \quad (\alpha_k)_{i+1/2,L}^{n+1/2} &= (\alpha_k)_i^n + (1 - (C_k)_i) \frac{(\Delta \alpha_k)_i}{2} \\ \text{else} \quad (\alpha_k)_{i+1/2,L}^{n+1/2} &= (\alpha_k)_i^n \end{aligned}$$

$$C_- = (u - a) \frac{\Delta t}{\Delta x}$$

$$C_0 = u \frac{\Delta t}{\Delta x}$$

$$C_+ = (u + a) \frac{\Delta t}{\Delta x}$$

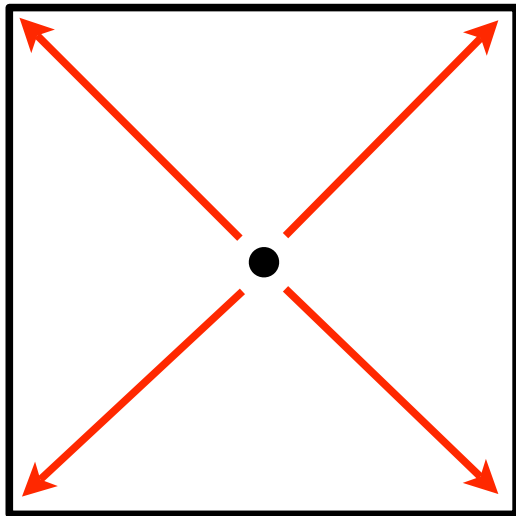
$$\begin{aligned} \text{If } (C_k)_{i+1} < 0 \quad (\alpha_k)_{i+1/2,R}^{n+1/2} &= (\alpha_k)_{i+1}^n - (1 + (C_k)_{i+1}) \frac{(\Delta \alpha_k)_i}{2} \\ \text{else} \quad (\alpha_k)_{i+1/2,R}^{n+1/2} &= (\alpha_k)_{i+1}^n \end{aligned}$$

- Colella, P. and Woodward, P., "The Piecewise parabolic Method (PPM) for Gasdynamical Simulations", J. Comput. Phys., **54**, 174-201 (1984).

## 2D slope limiter for unsplit schemes

$$u_{i,j+1/2}^{n+1/2} = u_{i,j}^n - C_x \Delta_x u_{i,j} + (1 - C_y) \Delta_y u_{i,j}$$

$$u_{i+1/2,j}^{n+1/2} = u_{i,j}^n + (1 - C_x) \Delta_x u_{i,j} - C_y \Delta_y u_{i,j}$$



If 1D slope limiters are used, 2D schemes may become oscillatory.

Predicted states involve 2D neighboring cells.

2D *moncen* slope: corner values must be bounded by the 8 neighboring initial values.

Surech, Ambady, "Positivity Preserving Schemes in Multidimensions", SIAM J. Sci. Comput., **22**, 1184-1198 (2000).

## Beyond second order Godunov schemes ?

### Smooth regions of the flow

More efficient to go to higher order.

Spectral methods can show *exponential convergence*.

More flexible approaches: use *ultra-high-order* shock-capturing schemes: 4th order scheme, ENO, WENO, discontinuous Galerkin and discontinuous element methods

### Discontinuity in the flow

More efficient to refine the mesh, since higher order schemes drop to first order.

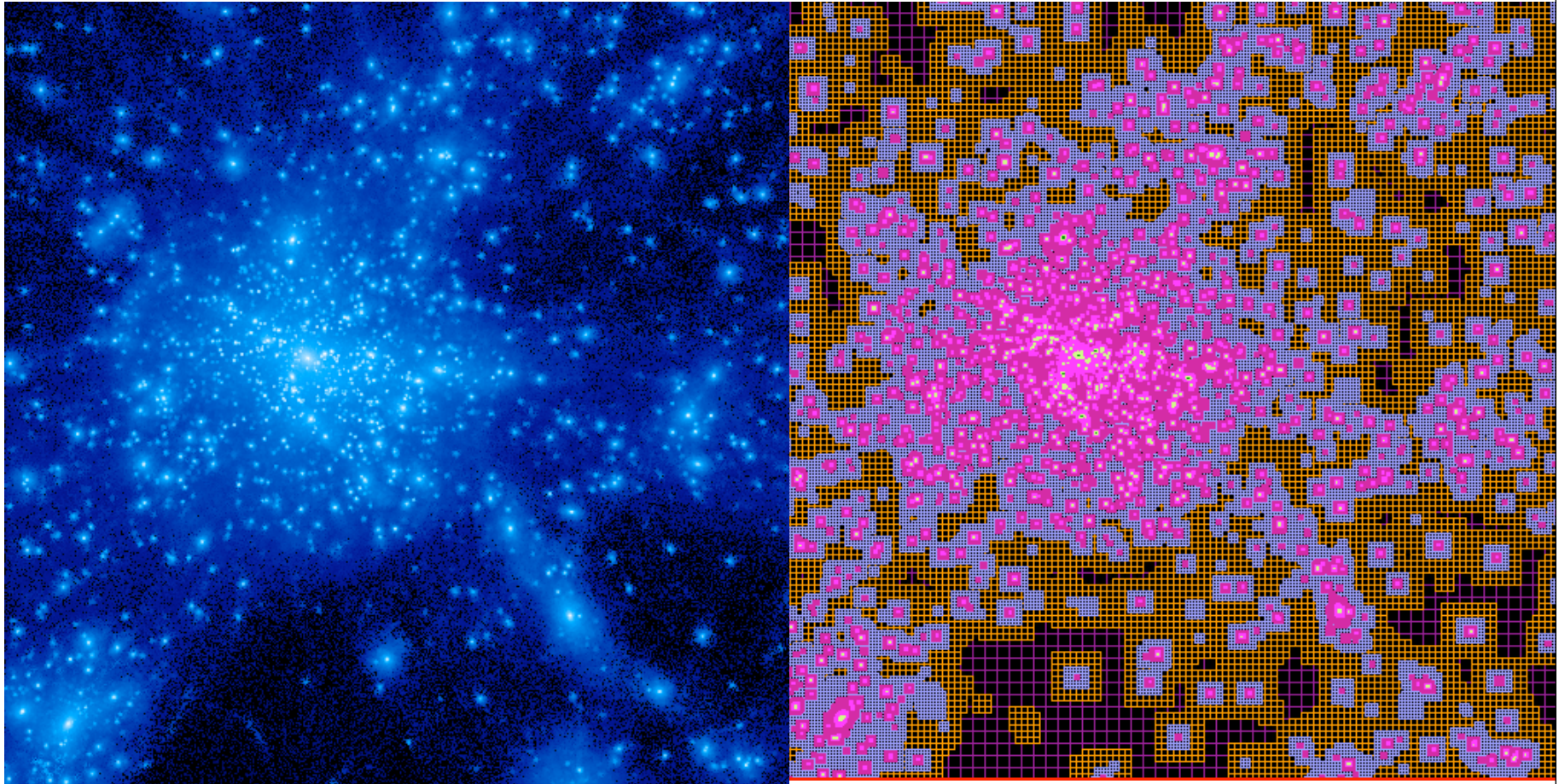
Adaptive Mesh Refinement is the most appealing approach.

### What about the future ?

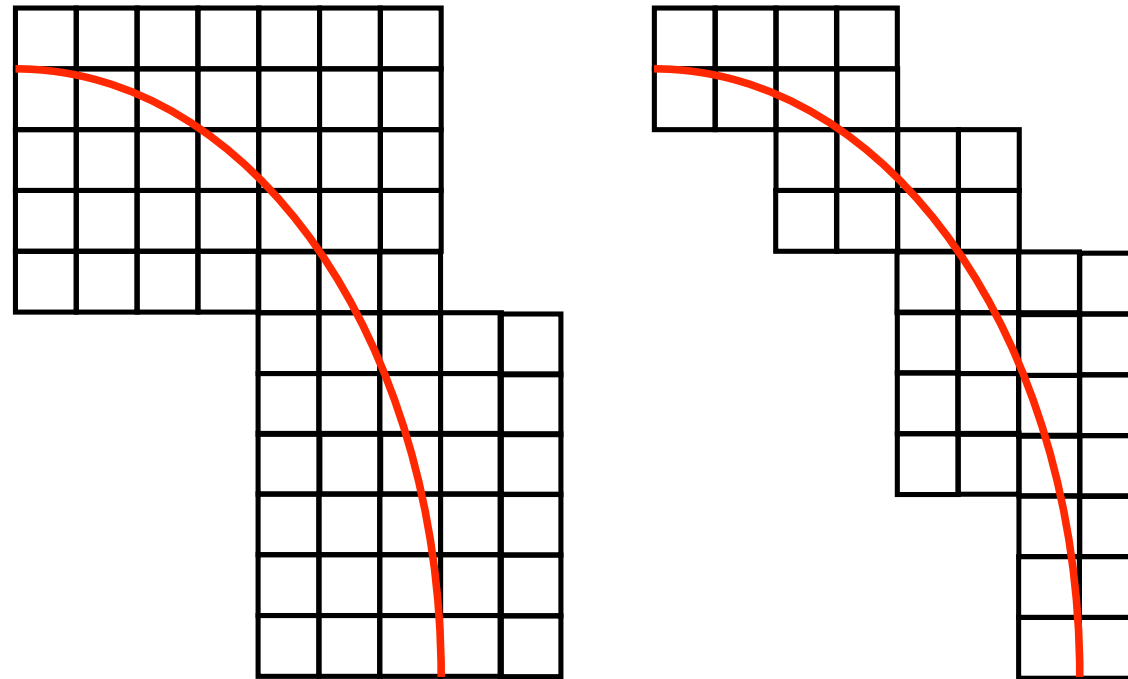
Combine the 2 approaches.

Usually referred to as "*h-p adaptivity*".

# Adaptive Mesh Refinement



## Patch-based versus tree-based





## A few AMR codes in astrophysics

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**ENZO:** Greg Bryan, Michael Norman...

**ART:** Andrey Kravtsov, Anatoly Klypin

**RAMSES:** Romain Teyssier

**NIRVANA:** Udo Ziegler

**AMRVAC:** Gabor Thot and Rony Keppens

**FLASH:** The Flash group (PARAMESH lib)

**ORION:** Richard Klein, Chris McKee, Phil Colella

**PLUTO:** Andrea Mignone (CHOMBO lib, Phil Colella)

**CHARM:** Francesco Miniati (CHOMBO lib, Phil Colella)

**ASTROBear:** Adam Frank...

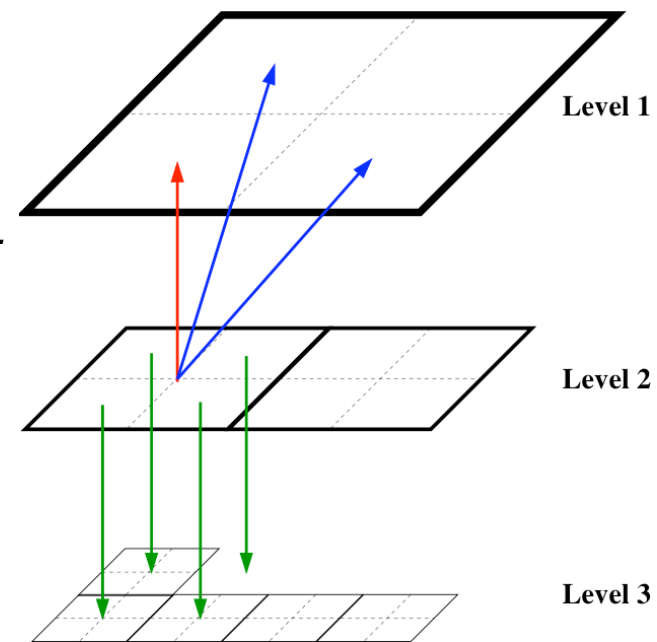
# Graded Octree structure

Fully Threaded Tree (Khokhlov 98).  
Cartesian mesh refined on a *cell by cell basis*.

**octs**: small grid of 8 cells

Pointers (arrays of index)

- 1 parent cell
- 6 neighboring parent cells
- 8 children octs
- 2 linked list indices



Cell-centered variables are updated level by level using linked lists.

Cost = 2 integer per cell.

Optimize mesh adaptation to complex flow geometries, but CPU overhead compared to unigrid can be as large as 50%.

2 type of cell: - “leaf” or active cell  
- “split” or inactive cell



# Refinement rules for graded octree

Compute the refinement map: flag = 0 or 1

## Step 1: mesh consistency

if a split cell contains at least one split or marked cell, then mark the cell with flag = 1 and mark its 26 neighbors

## Step 2: physical criteria

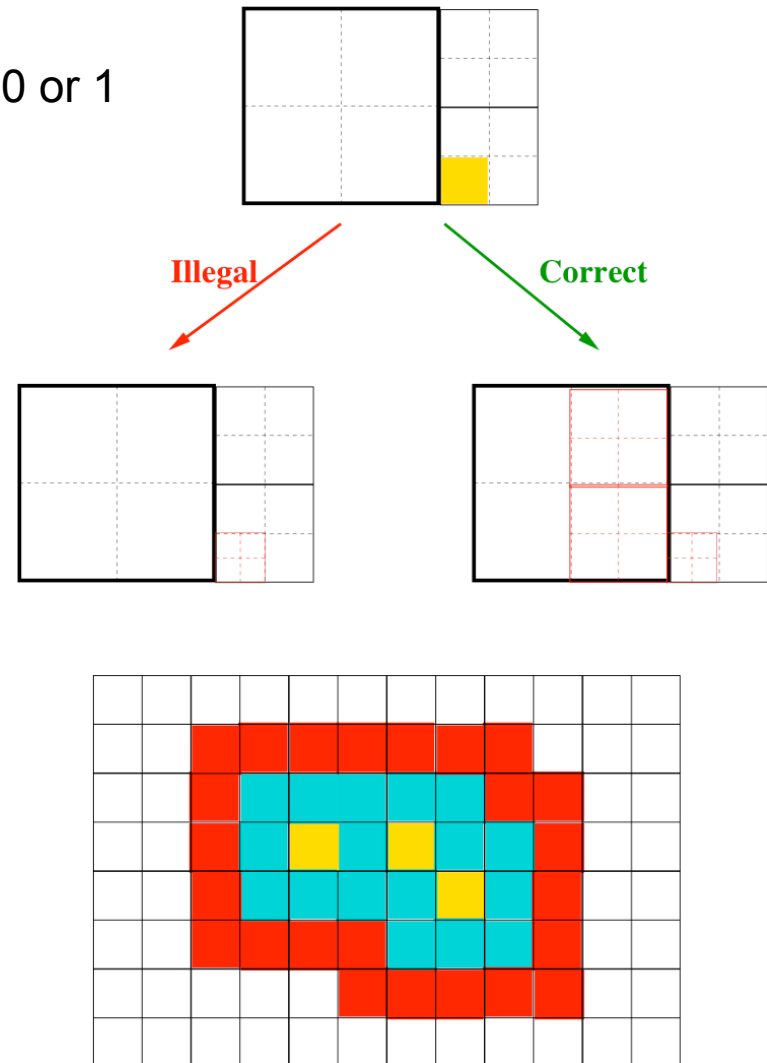
quasi-Lagrangian evolution, Jeans mass

geometrical constraints (zoom)

Truncation errors, density gradients...

## Step 3: mesh smoothing

apply a dilatation operator (mathematical morphology) to regions marked for refinement  $\rightarrow$  convex hull



# Godunov schemes and AMR

Berger & Oliger (84), Berger & Collela (89)

Prolongation (interpolation) to finer levels

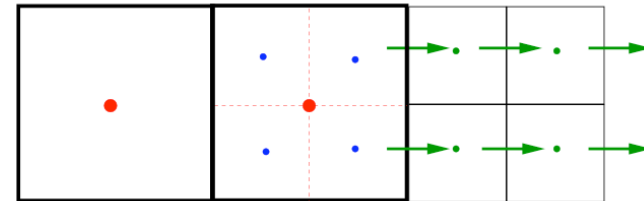
- fill buffer cells (boundary conditions)
- create new cells (refinements)

Restriction (averaging) to coarser levels

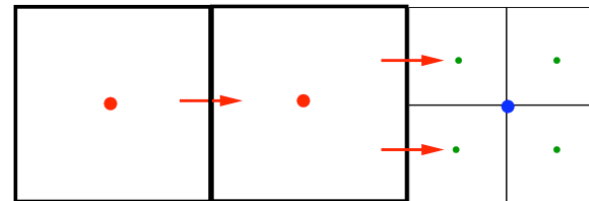
- destroy old cells (de-refinements)

Flux correction at level boundary

Solve for fine fluxes using buffer regions



Coarse flux: time and space average of fine fluxes



$$(\mathbf{F}_{i+1/2,j}^{n+1/2,\ell}) = \frac{(\mathbf{F}_{i+1/2,j-1/4}^{n+1/2,\ell+1}) + (\mathbf{F}_{i+1/2,j+1/4}^{n+1/2,\ell+1})}{2}$$

Careful choice of interpolation variables (conservative or not ?)

Several interpolation strategies (with  $\mathbf{R}^T \mathbf{P} = \mathbf{I}$ ) :

- straight injection
- tri-linear, tri-parabolic reconstruction

## Godunov schemes and AMR

Buffer cells provide boundary conditions for the underlying numerical scheme. The number of required buffer cells depends on the kernel of the chosen numerical method. *The kernel is the ensemble of cells on the grid on which the solution depends.*

- First Order Godunov: 1 cell in each direction

$$u_i^{n+1} = u_i^n(1 - C) + u_{i-1}^n C$$

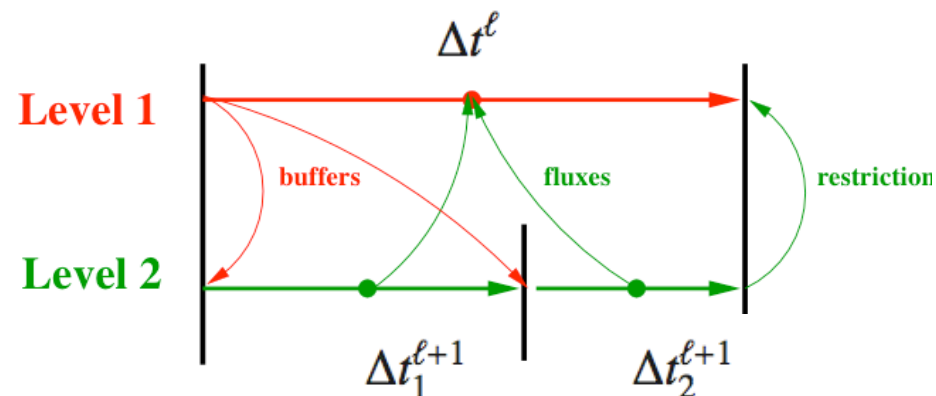
- Second order MUSCL: 2 cells in each direction

$$u_i^{n+1} = u_i^n(1 - C) + u_{i-1}^n C - \frac{C}{2}(1 - C)(\Delta u_i - \Delta u_{i-1}) = 0$$

- Runge-Kutta or PPM: 3 cells in each direction

Simple octree AMR requires 2 cells maximum. For higher-order schemes (WENO), we need to have a different data structure (patch-based AMR or augmented octree AMR).

# Adaptive Time Stepping

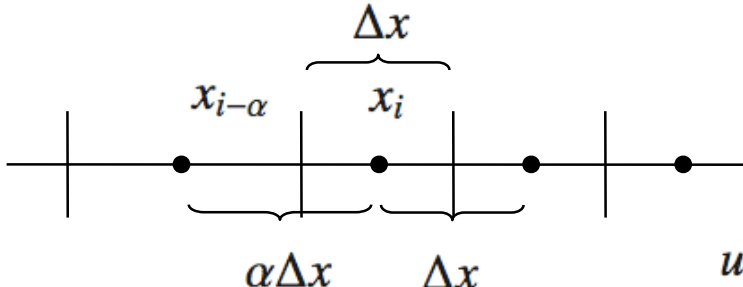


Time integration: single time step or recursive sub-cycling

- froze coarse level during fine level solves (one order of accuracy down !)
- average fluxes in time at coarse fine boundaries

$$(\mathbf{F}_{i+1/2,j}^{n+1/2,\ell}) = \frac{1}{\Delta t_1^{\ell+1} + \Delta t_2^{\ell+1}} \left( \Delta t_1^{\ell+1} \frac{(\mathbf{F}_{i+1/2,j-1/4}^{n+1/4,\ell+1}) + (\mathbf{F}_{i+1/2,j+1/4}^{n+1/4,\ell+1})}{2} + \Delta t_2^{\ell+1} \frac{(\mathbf{F}_{i+1/2,j-1/4}^{n+3/4,\ell+1}) + (\mathbf{F}_{i+1/2,j+1/4}^{n+3/4,\ell+1})}{2} \right)$$

## The AMR catastrophe



Assume  $a$  and  $C > 0$ .

$$u_{i+1/2}^{n+1/2} = u_i^n + (1 - C) \frac{\Delta x}{2} \left( \frac{\partial u}{\partial x} \right)_i$$

$$u_{i-1/2}^{n+1/2} = u_{i-\alpha}^n + (2\alpha - 1 - C) \frac{\Delta x}{2} \left( \frac{\partial u}{\partial x} \right)_{i-1}$$

$$\frac{u_i^{n+1} - u_i^n}{\Delta t} + a \frac{u_{i+1/2}^{n+1/2} - u_{i-1/2}^{n+1/2}}{\Delta x} = 0$$

First order scheme:  $\left( \frac{\partial u}{\partial t} \right) + \alpha a \left( \frac{\partial u}{\partial x} \right) = a \frac{\Delta x}{2} (\alpha^2 - C) \left( \frac{\partial^2 u}{\partial x^2} \right) + O(\Delta t^2, \Delta x^2)$

Second order scheme:  $\left( \frac{\partial u}{\partial t} \right) + a \left( \frac{\partial u}{\partial x} \right) = a \frac{\Delta x}{2} (\alpha - C)(1 - \alpha) \left( \frac{\partial^2 u}{\partial x^2} \right) + O(\Delta t^2, \Delta x^2)$

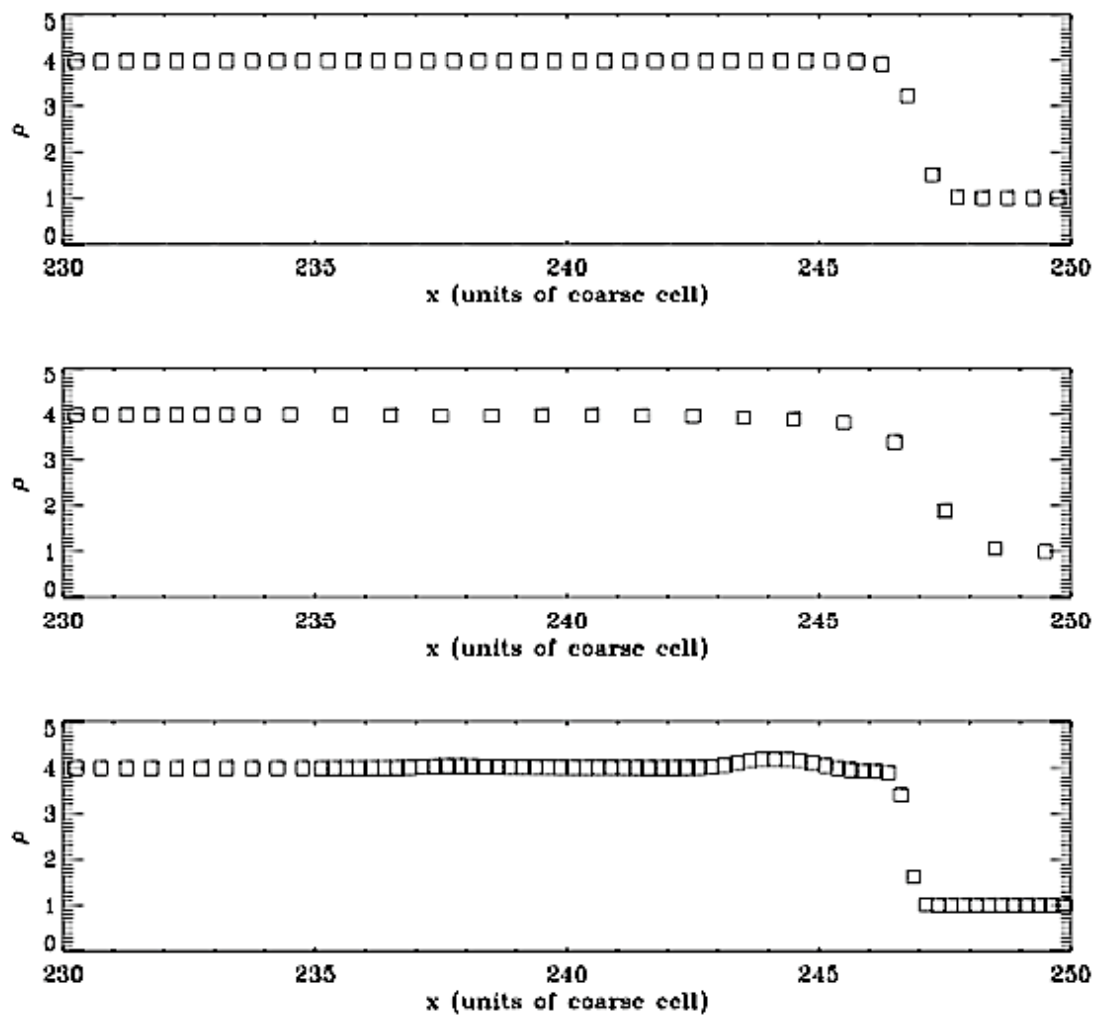
At level boundary, we loose one order of accuracy in the modified equation.

First order scheme: the AMR extension is *not consistent* at level boundary.

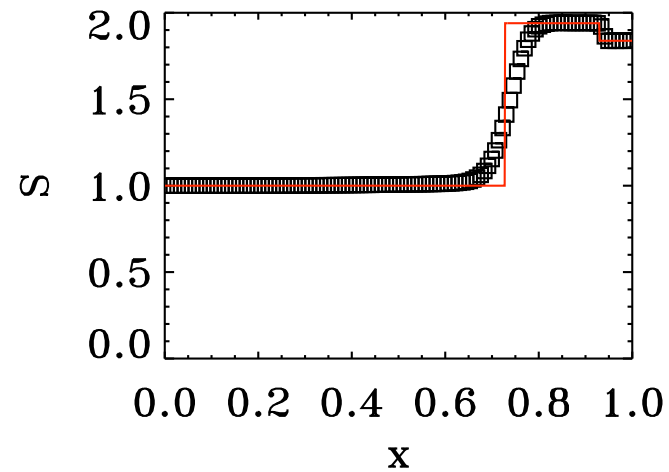
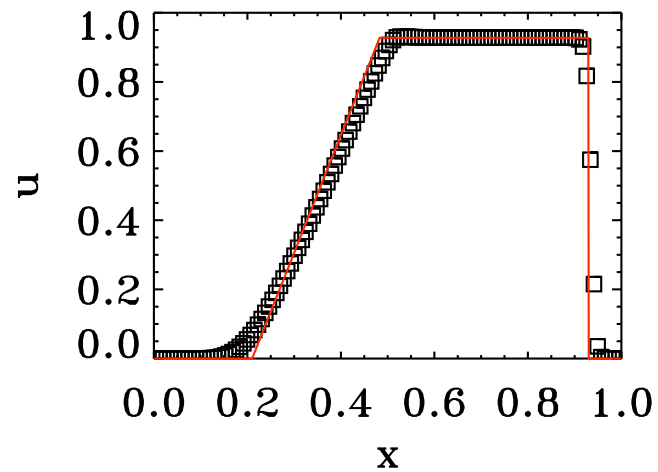
Second order scheme: for  $\alpha=1.5$ , AMR is *unstable* at level boundary.

Solutions: 1- refine gradients, 2- enforce first order, 3- add artificial diffusion

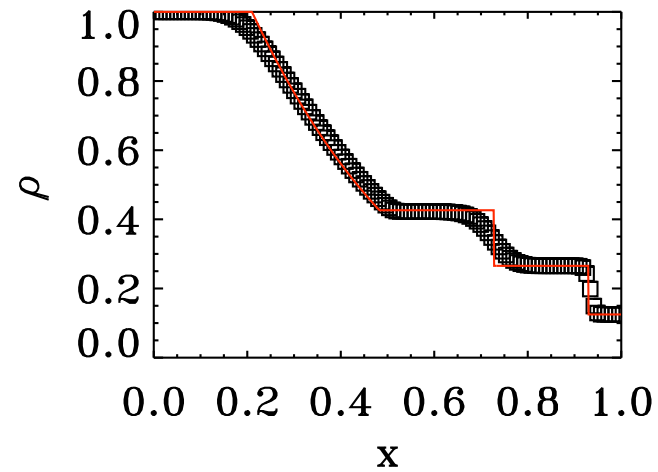
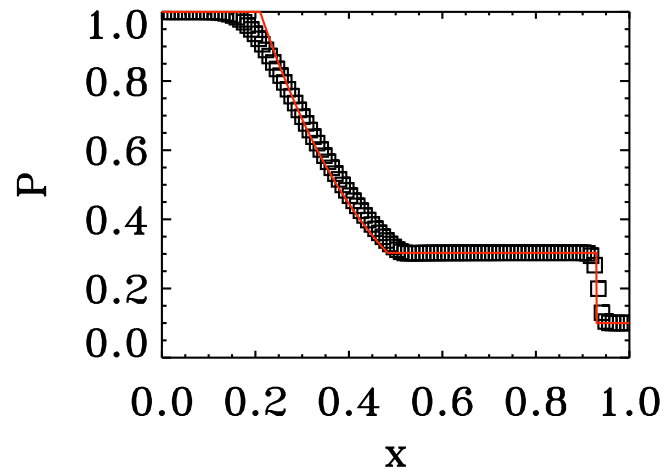
# Shock wave propagating through level boundary



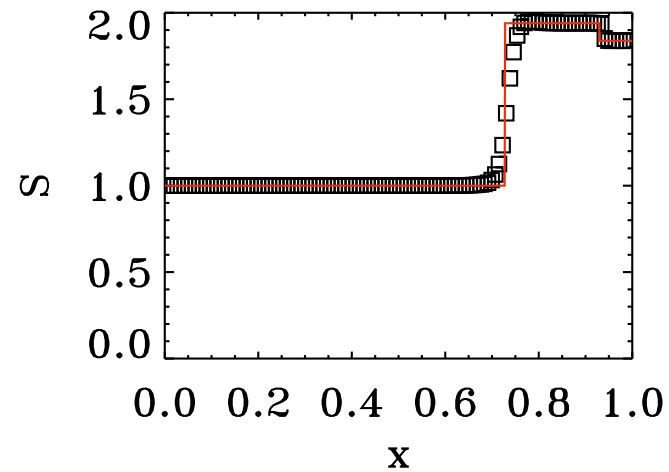
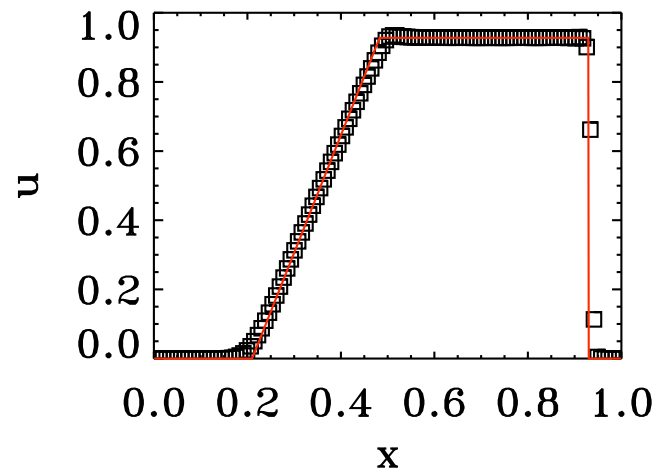
## Sod test with HLLC first order



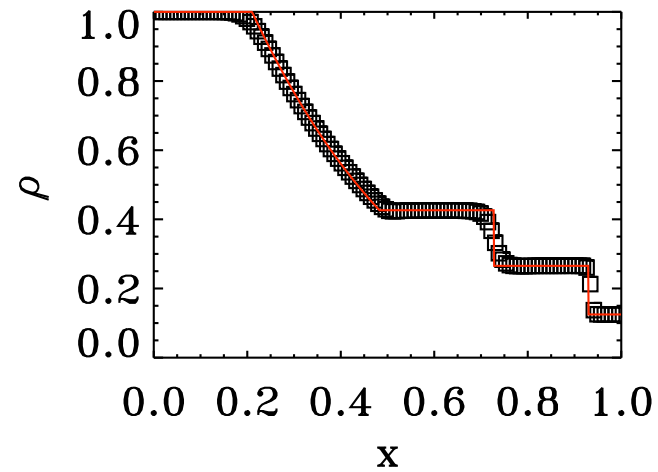
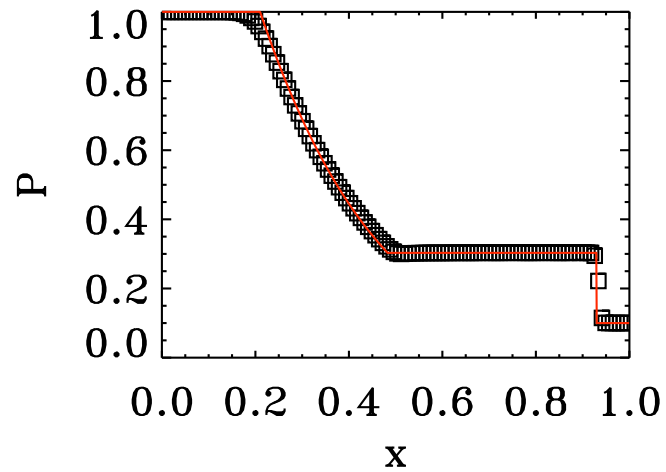
128 cells



## Sod test with HLLC and MinMod

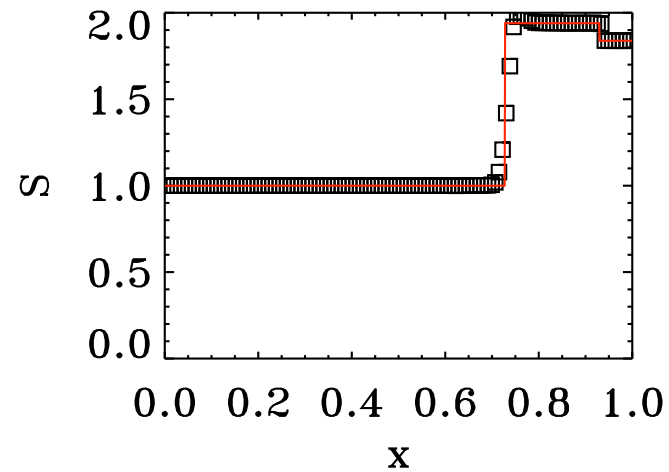
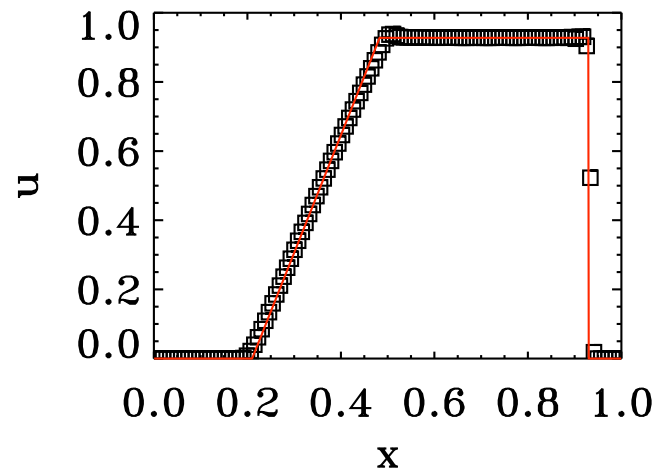


128 cells

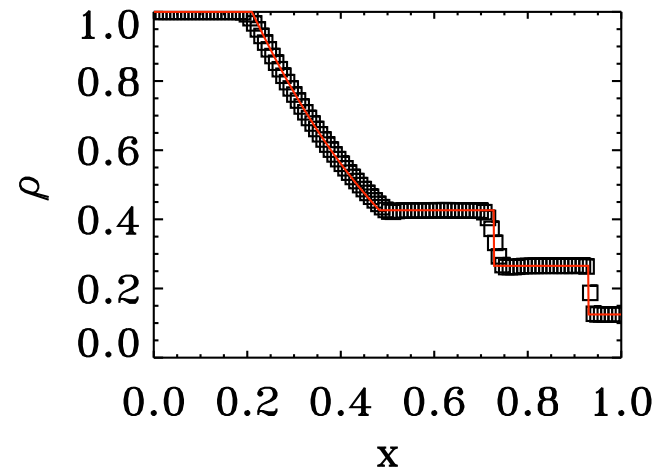
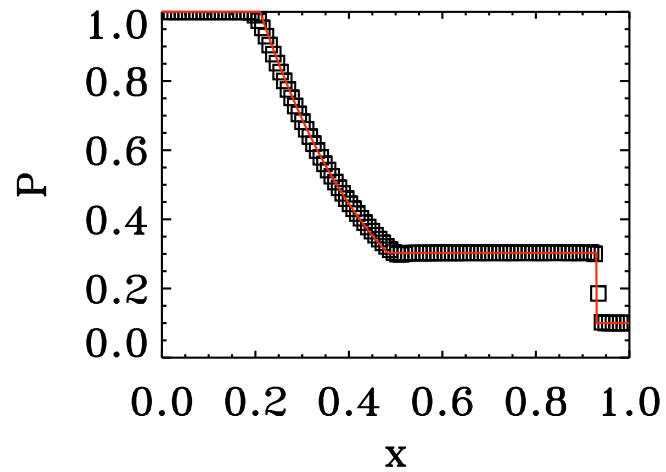




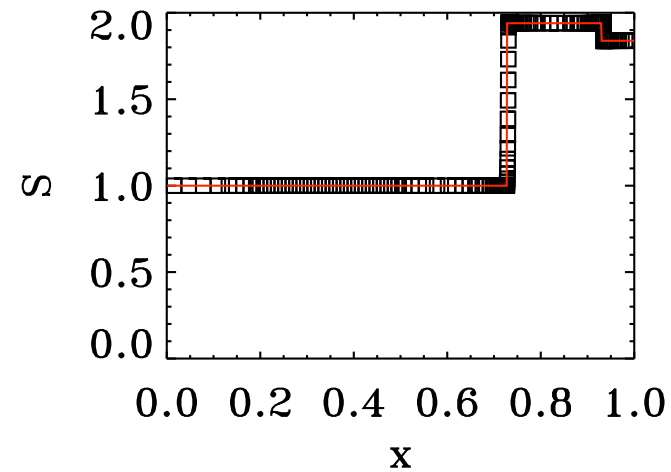
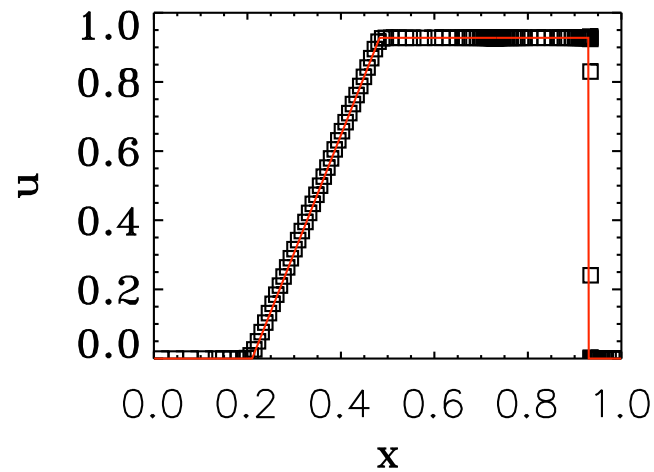
## Sod test with HLLC and MonCen



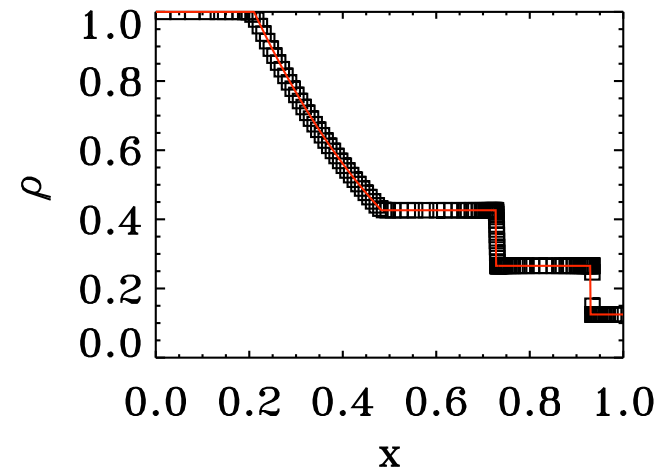
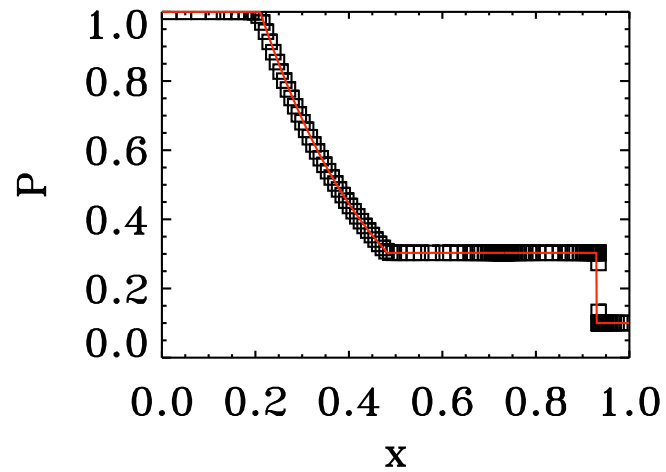
128 cells



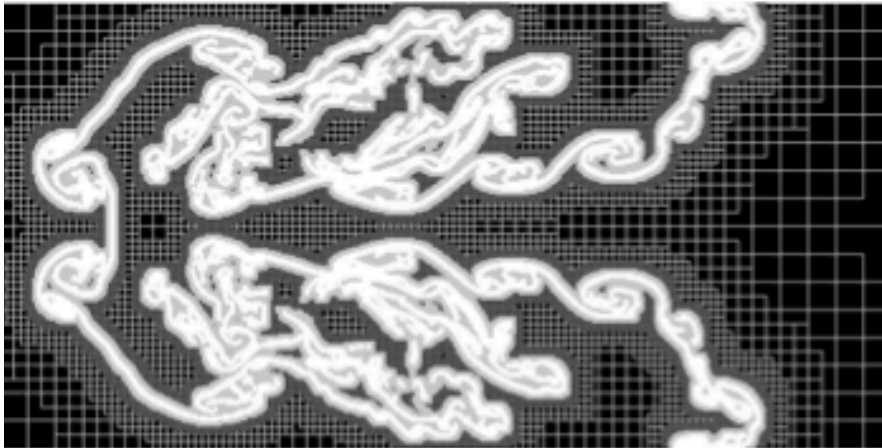
## Sod test with HLLC and AMR



153 cells

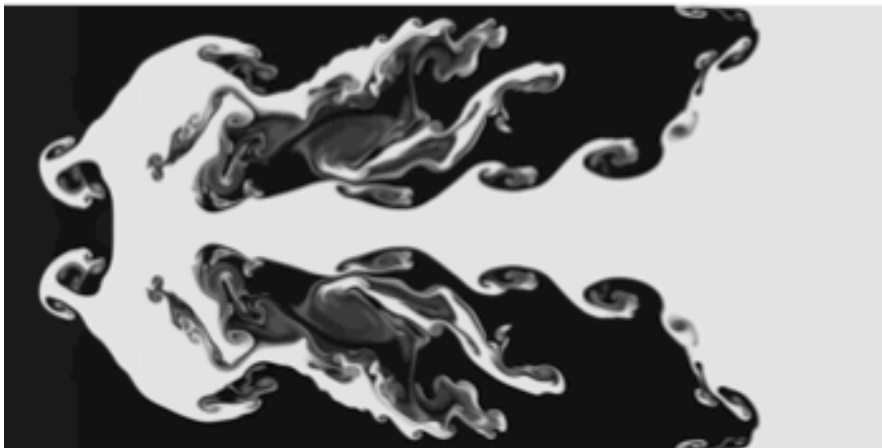


## Complex geometry with AMR



Maximum numerical dissipation occurs at the 2 fluids interface.

The optimal refinement strategy is based on density gradients.



The number of required cells is directly related to the *fractal exponent*  $n$  of the 2D surface.

$$N_{cell} \propto (\Delta x)^{-n}$$

# Cosmology with AMR

## Particle-Mesh on AMR grids:

Cloud size equal to the local mesh spacing

## Poisson solver on the AMR grid

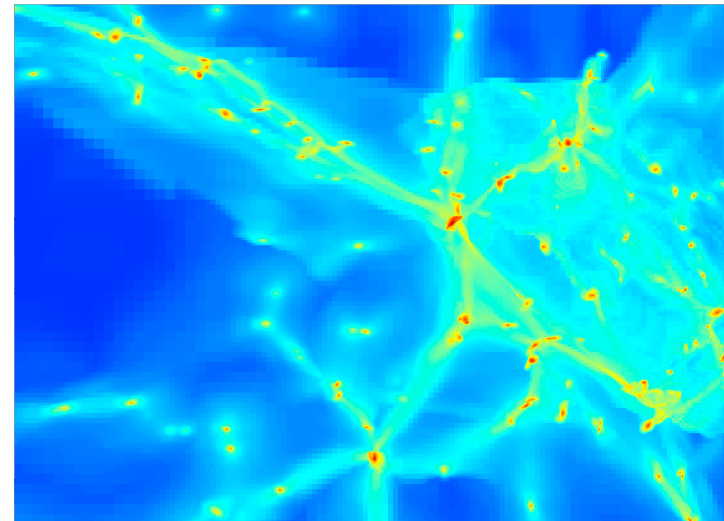
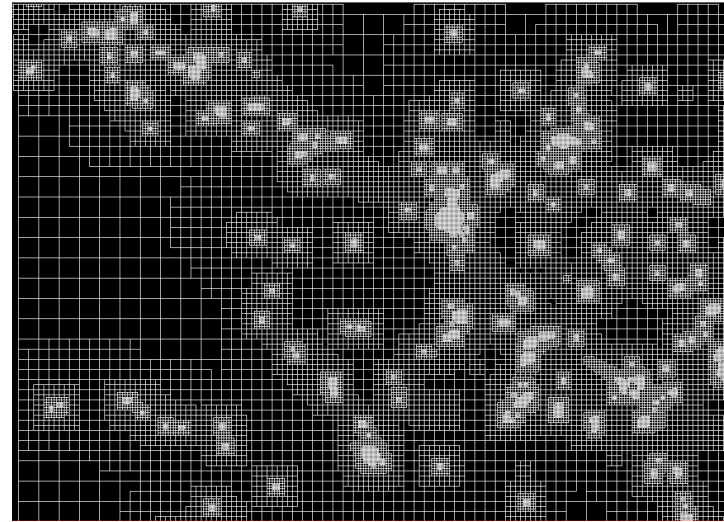
Multigrid or Conjugate Gradient  
Interpolation to get Dirichlet boundary conditions (one way interface)

## Quasi-Lagrangian mesh evolution:

roughly constant number of particles per cell

$$n = \frac{\rho_{DM}}{m_{DM}} + \frac{\rho_{gas}}{m_{gas}} + \frac{\rho_*}{m_*}$$

Trigger new refinement when  $n > 10$ -40 particles. The fractal dimension is close to 1.5 at large scale (filaments) and is less than 1 at small scales (clumps).



## RAMSES: a parallel graded octree AMR

- Tree-based AMR (octree structure) : the cartesian mesh is recursively refined *on a cell by cell basis*.
- Full connectivity : each “oct” have direct access to neighboring parent cells and to children “octs”. (memory overhead : 2 integers per cell).
- Optimize the mesh adaptivity to complex geometries, but CPU overhead can be as large as 50%.      Code is freely available [http://irfu.cea.fr/Projets/Site\\_ramses](http://irfu.cea.fr/Projets/Site_ramses)

**N body module :**      Particle-Mesh method on AMR grids (similar to the ART code).  
Poisson equation solved using Conjugate Gradient and Multigrid.

**Hydro module :**      *Unsplit* second order Godunov method : Riemann solver with  
piecewise linear reconstruction (option : MUSCL or PLMDE).

**Time integration :**      Single time step or W cycle (fine levels subcycling)

**Other**      Cooling & UV heating, Zoom simulation technology  
MPI based parallel implementation → *Space Filling Curves*

## Conclusion

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- Second-order in space and time with predictor-corrector: MUSCL
- Dispersive error term in the Modified Equation
- MinMod and MonCen slope limiters (+ characteristics tracing ?)
- 2D slope limiting for unsplit 2D schemes
- Patch-based versus Tree-based AMR
- AMR loses one order of accuracy at level boundary
- Refinement strategy and  $h/p$  adaptivity ?

**Next lecture: Hands on RAMSES**